

Mg
8.5 Al
0.5 Zn

AZ 80 A

- 1. GENERAL
This heat treatable magnesium-aluminum-zinc alloy is one of the strongest alloys for forgings and solid extrusions for service up to 300 F. It is available in various conditions. It has limited ductility at room temperature. It is difficult to forge and its use, therefore, limited to press forging at a slow rate. Welds in this alloy require stress relief.
- 1.01 Commercial Designation. AZ 80 A.
- 1.02 Alternate Designations. AZ 80 X, Dowmetal 0-1, Mazlo AMC 58 S (all obsolete).
- 1.03 Specifications. Table 1.03.

TABLE 1.03

AMS	Form	Federal
4360 C	Forgings (T5 Cond)	QQ - M - 40

- 1.04 Composition. Table 1.04.

TABLE 1.04

Source	AMS (1)	
	Min	Max
Aluminum	7.8	9.2
Zinc	0.20	0.8
Manganese	0.12 (a)	-
Silicon	-	0.30 (b)
Copper	-	0.05
Nickel	-	0.005
Iron	-	0.005
Other impurities (total)	-	0.30
Magnesium	Balance	

(a) Dow gives 0.15
(b) Dow gives 0.05

- 1.07 Forms and Conditions Available
- 1.071 Solid extrusions and forgings are available in all commercial sizes for magnesium alloys.
- 1.072 Solid extrusions are available in the F and T5 Conditions.
- 1.073 Forgings are available in the F, T5 and T6 Conditions.
- 1.08 Melting and Casting Practice. See AZ 31 B.
- 1.09 Special Considerations. See AZ 31 B also.
- 1.091 Stress relief is required after forming and welding to prevent stress cracking.
- 1.092 Forging this alloy is difficult.

2. PHYSICAL AND CHEMICAL PROPERTIES

- 2.01 Thermal Properties
- 2.011 Melting range. 800 to 1115 F.
- 2.012 Phase changes. Alloy is subject to precipitation.
- 2.013 Thermal conductivity. 29.0 Btu ft per (hr sq ft F).
- 2.014 Thermal expansion. See AZ 31 B.
- 2.015 Specific heat. 0.25 Btu per (lb F).
- 2.016 Thermal diffusivity. 1.08 sq ft per hr.
- 2.02 Other Physical Properties
- 2.021 Density. 0.0649 lb per cu in. 1.80 gr per cu cm.
- 2.022 Electrical resistivity. 5.71 microhm in.
- 2.023 Magnetic properties. Alloy is nonmagnetic.
- 2.03 Chemical Properties. Same as AZ 31 B.
- 2.04 Nuclear Properties. See AZ 31 B.

3. MECHANICAL PROPERTIES

- 3.01 Specified Mechanical Properties
- 3.011 AMS and producers' specified mechanical properties, Table 3.011.

TABLE 3.011

Source	(2)										AMS (1)				
	AZ 80 A														
	Extrusions										Forgings		Die Forgings		Hand Forgings
Form	F								T5		F	T5	T5		TS
Condition	F				T5				F	T5	T5		TS		
Thickness - in	Under 0.250	0.250 to 1.499	1.500 to 2.499	2.500 to 5.000	Under 0.250	0.250 to 1.499	1.500 to 2.499	2.500 to 5.000	-	-	< 2	≥ 2	≥ 50lb	-	
											< \$0.1b	(a)	(b)		
F _{tu} , min - ksi	43	43	43	42	47	48	48	45	42	42	42	34	40	42	
F _{ty} , min - ksi	28	28	28	27	30	33	33	30	26	28	30	22	26	28	
e, min - percent	9	9	8	4	5	4	4	2	5	2	2	2	3	2	
F _{cy} , min - ksi	-	17	17	17	-	28	27	26	18	25	-	-	-	-	
F _{su} , min - ksi	19	19	19	19	-	21	21	-	20	20	-	-	-	-	
F _{bu} , min - ksi	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
(e/D=1.5)	48	48	48	48	-	-	-	-	-	50	-	-	-	-	
(e/D=2.0)	56	56	56	56	-	-	-	-	-	70	-	-	-	-	
F _{by} , min - ksi	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
(e/D=1.5)	36	36	36	36	-	-	-	-	-	42	-	-	-	-	
(e/D=2.0)	40	40	40	40	-	-	-	-	-	50	-	-	-	-	
Hardness (c)															
RHN, min	-	-	-	-	-	-	-	-	-	-	-	-	82	70	

(a) Individual Specimen
(b) Average of 8 Specimen
(c) 1000 Kg 10mm.

- 1.05 Heat Treatment.
- 1.051 Stress relief after forming or welding.
- 1.0511 F Condition. 500 F, 15 min.
- 1.0512 T5 Condition. 400 F, 1 hr.
- 1.0513 T6 Condition. 600 F, 1/4 hr.
- 1.052 Solution treat forgings to T6 Condition. 750 F, 2 to 4 hr, air cool.
- 1.053 Age F Condition to T5 Condition. 350 F, 24 hr.
- 1.054 Age T4 Condition of forgings to T6 Condition. 350 F, 24 hr.
- 1.06 Hardenability. Alloy hardens fully in all section sizes on aging either hot worked or solution treated material.

- 3.02 Mechanical Properties at Room Temperature. See 3.03 also.
- 3.021 Typical mechanical properties, Table 3.021.

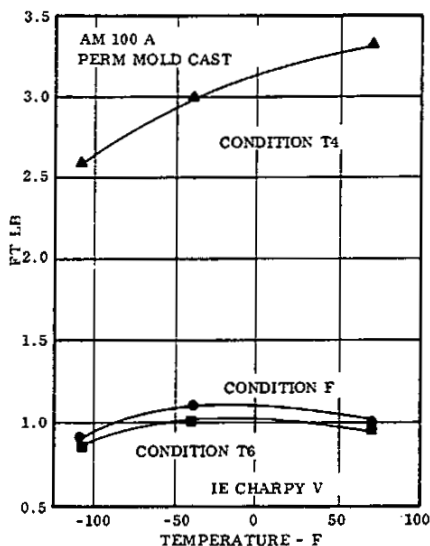


FIG. 3.0331 EFFECT OF LOW TEMPERATURES ON CHARPY V-NOTCH IMPACT ENERGY. (13)

	Mg
IO	Al
O.I	Mn

AM 100 A

REFERENCES

- Gantz, M. E., Cnigerick, E. M., and Wood, R. T., "Magnesium Alloy Permanent Mold and Semi-Permanent Mold Castings," Transactions AFS, Volume 61 (1953) pp. 502-509.
- AMS 4455A (May 1, 1968)
- AMS 4483A (May 1, 1968)
- "Heat Treatment of Magnesium Alloys," ASM Metals Handbook, 8th edition, Volume 2 (1964) pp. 292-297.
- "Heat Treating Sand and Permanent Mold Magnesium"
- "Magnesium Technical Service Repair Manual for Aircraft Structures of Magnesium Sheet, Extrusions, Forgings, and Castings," The Dow Chemical Company, Metal Products Department (June 23, 1967).
- Castings, "Form 141-35-68, The Dow Chemical Company, Metal Products Department (June 23, 1968).
- "Crucible Melting of Magnesium Alloys," Bulletin No. 181-27, The Dow Chemical Company, Magnesium Sales Department.
- Recommended Practices for Sand Casting Aluminum and Magnesium Alloys, book published by the American Foundrymen's Society, 2nd edition (1965).
- Stonebrook, E. E., "Solidification and Heat Treatment of Aluminum and Magnesium Alloys," Foundry, Volume 87 (September 1959) pp. 90-101.
- Holdeman, G. E., "Metallography in the Magnesium Foundry," Transactions AFS, Volume 64 (1956) pp. 698-708.
- George, P. F., "Metallography of Cast Magnesium Alloys," Transactions AFS, Volume 57 (1949) pp. 133-148.
- Cast Metals Handbook, published by the American Foundrymen's Society, 4th edition (1957) pp. 280-284.
- "Designing with Magnesium," published by the American Magnesium Corporation, subsidiary of Alcoa (1945).
- "Properties of Magnesium Alloys," ASM Metals Handbook, 8th edition, Volume I (1961) pp. 1095-1112.
- Hallowell, J. B., and Ogden, H. R., "An Introduction to Magnesium Alloys," DMIC Report 206 (August 26, 1964) Battelle Memorial Institute, Columbus, Ohio.
- Jensen, J. W., "Damping Capacity-- Causes and Effects," Light Metal Age, Volume 22 (December 1964) pp. 4-8.
- Hanawalt, J. D., Nelson, C. E., and Peloubet, J. A., "Corrosion Studies of Magnesium and Its Alloys," Metals Technology, Volume 8 TP No. 1353 (September 1941).
- Hanawalt, J. D., Nelson, C. E., and Busk, R. S., "Properties and Characteristics of Common Magnesium Casting Alloys," Transactions AFS, Volume 53 (1945) pp. 77-86.
- "Corrosion of Magnesium Alloys," ASM Metals Handbook, Volume I (1961) pp. 1086-1094.
- Herrick, K., "Magnesium-Base Alloys, Investment Cast Properties," Transactions AFS, Volume 69 (1961) pp. 179-188.
- "Selection and Application of Magnesium and Magnesium Alloys," ASM Metals Handbook, Volume I (1961) pp. 1067-1086.
- "Arc Welding Magnesium," Form No. 141-300-67, The Dow Chemical Company, Metal Products Department (1965).
- "Resistance Welding Magnesium," Form No. 141-304-66, The Dow Chemical Company, Metal Products Department (1966).
- "Magnesium Finishing," Form No. 141-288-67, The Dow Chemical Company, Metal Products Department (1967).

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TABLE 3.021

Source	(2)(3)							
	AZ 80 A							
Form	Extrusions				Forgings			
Condition	F				T5			
Thickness - in	< 1.500	1.500 to 5.000	5.000 to 10.000	> 10.000	< 1.500	1.500 to 5.000	5.000 to 10.000	> 10.000
F _{tu} - ksi	49	48	46	44	55	53 to 50	50	50
F _{ty} - ksi	36	35	31	28	38 to 40	39 to 38	34	34
e, percent	11	9	8	7	7	6	6	5
F _{cy} - ksi	-	-	25	25	35	32 to 31	28	27
F _{su} - ksi	22	-	22	-	24	-	23	-
F _{ru} - ksi (e/D = 2.0)	68	-	-	-	-	-	80	-
F _{br} - ksi (e/D = 2.0)	68	-	-	-	-	-	-	-

- 3.022 Stress strain curves in tension and compression for forgings, Fig. 3.022.
- 3.023 Effect of exposure to elevated temperatures on tensile properties of extrusions, Fig. 3.023.
- 3.024 Effect of stress concentration on notch strength ratio of extrusions, Fig. 3.024.
- 3.03 Mechanical Properties at Various Temperatures
- 3.031 Short time tension properties
- 3.0311 Effect of exposure and test temperature on tensile properties of extrusions and forgings, Fig. 3.0311.
- 3.032 Short time properties other than tension
- 3.033 Static stress concentration effects
- 3.04 Creep and Creep Rupture Properties
- 3.041 Total strain curves for forgings at 200 and 300 F, Fig. 3.041.
- 3.05 Fatigue Properties
- 3.051 Fatigue properties of extrusions and forgings, Table 3.051.

- 4.012 Forming of extrusions in the T5 Condition can be best performed at 380 F, with the time at temperature limited to 1 hr.
- 4.013 Forging of this alloy is difficult and, therefore, limited to press forging simple designs. In addition, the rate of deformation should be low. Hammer forgings are not generally produced. Upset forging should be preceded by heating at 750 to 775 F, for 2 to 6 hr, to prevent hot shortness.
- 4.02 Machining. Same as AZ 31 B.
- 4.03 Welding. Extrusions and forgings can be welded in all conditions by the inert gas electric arc method. No flux is necessary. The preferred welding rod is AZ 92 A. Welded assemblies must be stress relieved to avoid cracking.
- 4.04 Heating and Heat Treating. See AZ 63 A.
- 4.05 Surface Treating. See AZ 63 A.

TABLE 3.051

Source	(5, p. 4.115, 4.116)								
	Form	Temp F	Method	(1) Extrusions		(2) Forgings			
				Stress Ratio	Stress Concentration	Fatigue Strength - ksi at Cycles			
Cond			A	R	10 ⁵	10 ⁶	10 ⁷	10 ⁸	
(1) F	RT	Rot beam	∞	-1	Smooth	25 to 30	23 to 28	21 to 26	20 to 24
(2) F					(K = 1)	28 to 30	24 to 26	20 to 22	18 to 20
(2) T4						-	21 to 24	18 to 21	16 to 18
(2) T5						26 to 30	22 to 25	19 to 21	16 to 19
(2) T6						23 to 27	19 to 22	16 to 19	14 to 16
(1) F	RT	Rev bend	∞	-1	As extruded	19 to 21	13 to 16	12 to 14	-
(2) T6					As forged	16 to 20	13 to 16	12 to 15	-
Source	(6)								
(1) T5	RT	Rev bend	∞	-1	Smooth	20	15	12.5	-
(2) T5		Rot beam	∞	-1	Smooth	25	-	20	18
					Notched	16.5	-	9.5	9.5
					K = 2	20.5	-	11	10.5
					Notched	10.5	-	6.5	5
					K = 2				

- 3.06 Elastic Properties
- 3.061 Modulus of elasticity, 6500 ksi.
- 3.062 Modulus of rigidity, 2400 ksi.
- 3.063 Tangent modulus curves in tension and compression for extrusions in T5 Condition, Fig. 3.063.

4. FABRICATION

- 4.01 Forming and Casting
- 4.011 Forming of extrusions in the F Condition can be best performed at 550 F, with the time at temperature limited to 1/2 hr.

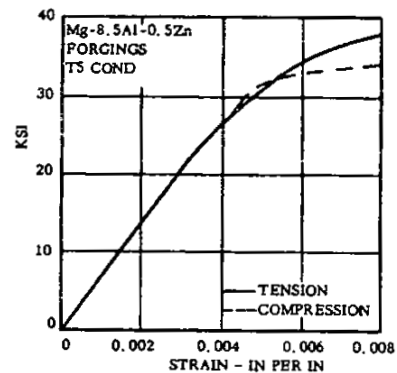


FIG. 3.022 STRESS STRAIN CURVES IN TENSION AND COMPRESSION FOR FORGINGS (2)

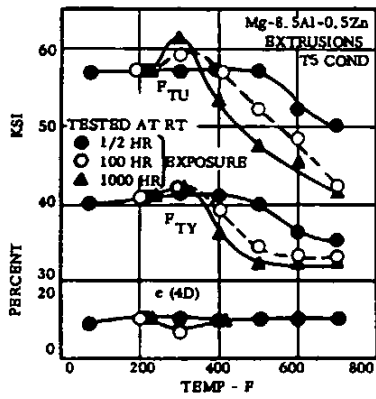
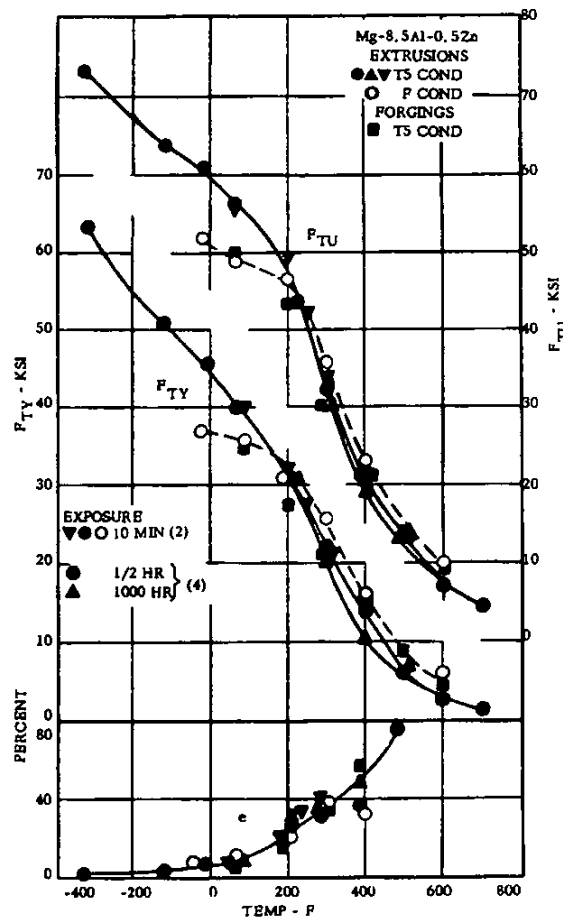


FIG. 3.023 EFFECT OF EXPOSURE TO ELEVATED TEMPERATURES ON TENSILE PROPERTIES OF EXTRUSIONS (4)



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FIG. 3.0311 EFFECT OF EXPOSURE AND TEST TEMPERATURE ON TENSILE PROPERTIES OF EXTRUSIONS AND FORGINGS (2)(4)

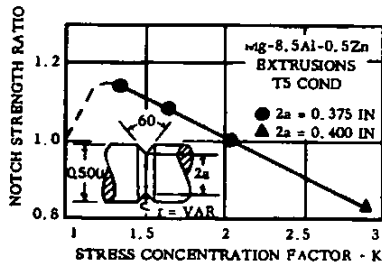


FIG. 3.024 EFFECT OF STRESS CONCENTRATION ON NOTCH STRENGTH RATIO OF EXTRUSIONS (2)

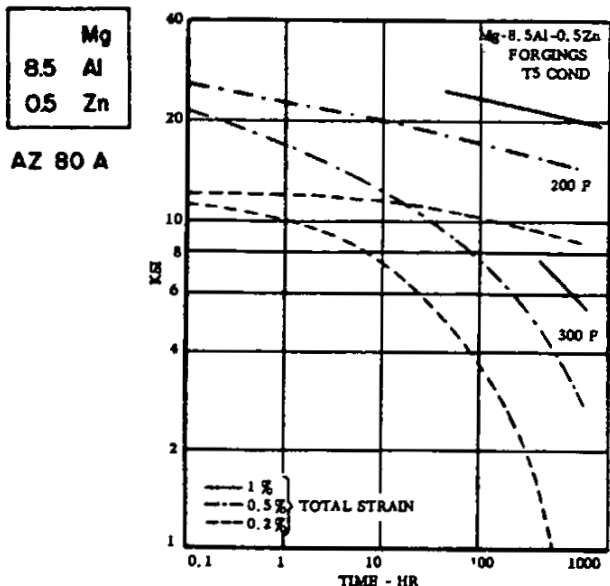


FIG. 3.041 TOTAL STRAIN CURVES FOR FORGINGS AT 200 AND 300 F (5)

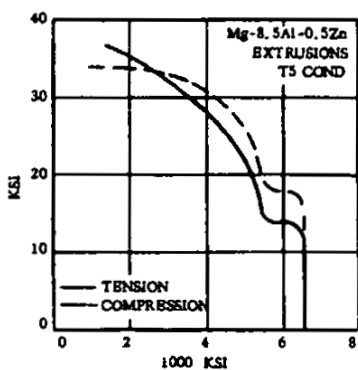


FIG. 3.063 TANGENT MODULUS CURVES IN TENSION AND COMPRESSION FOR EXTRUSIONS IN T3 CONDITION (5)

REFERENCES

- 1 AMS 4360 C, (June 15, 1952)
- 2 The Dow Chemical Co., "Magnesium Design," (1957)
- 3 The Dow Chemical Co., Magnesium Dept., "Magnesium Alloys and Products," (1958)
- 4 The Aluminum Co. of America, "Mechanical Properties at Various Temperatures for AZ 80 A," Alcoa Research Laboratory Data Sheet, (Aug. 7, 1957)
- 5 ANC-S, "Strength of Metal Aircraft Elements," (March 1955)
- 6 Ordnance Corps Pamphlet, "Ordnance Materials Handbook, Magnesium and Magnesium Alloys," ORDP 20-303, (Sept. 1956)