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## NONFERROUS ALLOYS

1. GENERAL  
This heat treatable magnesium alloy, which is usually used in either the T5 (artificially aged) or the T6 (solution treated and aged) condition is produced in the form of extrusions and forgings. It has similar high strength properties to AZ80A, and its ductility and toughness are superior to those of any other magnesium extrusion and forging alloys. At temperatures above 300 F, its strength falls below that of the thorium containing alloy, HM31A. ZK60A has poor arc-weldability, but reasonably good resistance-weldability. This alloy is used for highly stressed parts, primarily in aerospace and military applications at temperatures up to 300 F. A variation of this alloy, designated (P)ZK60B, was formerly made by extruding atomized ZK60A alloy into structural shapes. It was used commercially to a very limited extent but is now obsolete and no longer produced.
- 1.01 Commercial Designation  
ZK60A.
- 1.02 Alternate Designations  
ZK60 and ZW6(British).
- 1.03 Specifications  
Table 1.03
- 1.04 Composition  
Table 1.04.
- 1.05 Heat Treatment
- 1.051 T5 condition: age as-forged (F condition) material at 290 to 310 F 24 hours, air cool (6).
- 1.052 T6 condition: Solution treat at 925 to 935 F 2 hours in furnace atmosphere containing 1.0 percent minimum SO<sub>2</sub> or 5.0 percent minimum CO<sub>2</sub>, air cool, age at 290 to 310 F 24 hours, air cool (6)(7).
- 1.053 O condition, full anneal: 550 F 1 hour, air cool (6).
- 1.054 Stress relief after forming or welding: for material in F condition 500 F 15 minutes, air cool; for material in T5 or T6 condition 300 F 1 hour, air cool (6)(8).
- 1.06 Hardness  
Table 1.06.
- 1.07 Forms and Conditions Available  
Table 1.07.
- 1.08 Melting and Casting Practice  
(See AZ31B, Code 3601, Section 1.08.)
- 1.09 Special Considerations
- 1.091 In the T5 and T6 conditions, ZK60A loses room temperature strength rapidly as a result of exposures to temperatures above 300 F.
2. PHYSICAL AND CHEMICAL PROPERTIES
- 2.01 Thermal Properties
- 2.011 Melting range. 970 to 1175 F (10).
- 2.012 Phase changes. In the F condition, this alloy contains magnesium-zinc and zirconium-zinc compounds within an alpha magnesium matrix. T5 (artificial aging) treatment of the F material causes finely divided precipitation of more of these compounds. Solution treatment (930 F) causes most of the compounds to dissolve in the matrix, and artificial aging (about 300 F) causes them to reprecipitate in finely divided form. Prolonged aging at higher temperatures causes coagulation and growth of the precipitates resulting in deterioration of strength.
- 2.0121 Time-temperature-transformation diagrams.
- 2.013 Thermal conductivity, Figure 2.013.
- 2.014 Thermal expansion.  $15.0-15.1 \times 10^{-6}$  per F from 65 to 392 F (10).
- 2.015 Specific heat, Figure 2.015.
- 2.016 Thermal diffusivity. T5 condition, 2.59 sq ft per hr at 68 F (10).
- 2.02 Other Physical Properties
- 2.021 Density. 0.0660 lb per cu in (10).
- 2.022 Electrical properties, Table 2.022.
- 2.023 Magnetic properties, non-magnetic.
- 2.024 Emissance.
- 2.0241 Effect of temperature on normal total emissance of magnesium alloys in general, Figure 2.0241.
- 2.025 Damping capacity. At a stress equal to 0.1 F<sub>ty</sub> the specific damping capacity of ZK60A in both the F and T5 conditions is 0.2, which is a low level compared with other magnesium alloys (13).
- 2.03 Chemical Properties
- 2.031 Corrosion resistance.
- 2.0311 ZK60A, like other magnesium alloys, is subject to stress corrosion and to general and pitting corrosion in industrial, marine, and moist environments. With suitable surface treatment and painting, it performs satisfactorily in all types of natural environments with the exception of continuous immersion in water (14).
- 2.0312 Effect of stress on time to stress corrosion failure of specimens exposed to atmosphere in Washington, D.C., Figure 2.0312.
- 2.0313 For a discussion of galvanic corrosion, see AZ31B, Code 3601, Section 2.0312.
- 2.032 Safety precautions should be directed to the prevention of fires, burns, and explosions (see HZ32A, Code 3408, Section 2.032).
- 2.04 Nuclear Properties
3. MECHANICAL PROPERTIES
- 3.01 Specified Mechanical Properties  
Table 3.01.
- 3.02 Mechanical Properties at Room Temperature
- 3.021 Tension.
- 3.0211 Stress-strain diagrams.
- 3.02111 Stress-strain curves at room temperature for extrusions in different conditions, orientation, and sections sizes, Figure 3.02111.
- 3.0212 Tensile properties of extrusions and forgings in various conditions and sizes, Table 3.0212.
- 3.0213 Effect of exposures at elevated temperatures on tensile properties of extrusions at room temperature, Figure 3.0213.
- 3.0214 Tensile properties in various sections of aircraft wheel forging, Table 3.0214.
- 3.0215 Effect of rolling reduction and orientation on the tensile properties of roll forged rings produced directly from cast blanks, Figure 3.0215.
- 3.0216 Effect of rolling reduction and orientation on the tensile properties of roll forged rings produced from upset blanks, Table 3.0216.
- 3.022 Compression.
- 3.0221 Stress-strain diagrams.
- 3.02211 Compressive stress-strain curves at room temperature for extrusions in different conditions, orientations, and sections sizes, Figure 3.02211.
- 3.0222 Compressive yield strength of extrusions and forgings in various conditions and sizes, Table 3.0222.
- 3.0223 Effect of exposures at elevated temperatures on compressive yield strength at room temperature, Figure 3.0223.
- 3.0224 Effect of rolling reduction and orientation on the compressive yield strength of roll forged rings produced directly from cast blanks, Figure 3.0224.
- 3.0225 Effect of rolling reduction and orientation on the compressive yield strength of roll forged rings produced from upset blanks, Table 3.0225.
- 3.023 Impact (see Figure 3.0331).
- 3.024 Bending.
- 3.025 Torsion and shear.
- 3.0251 Shear strengths of extrusions and forgings in various conditions and sizes, Table 3.0251.
- 3.026 Bearing.
- 3.0261 Bearing properties of extrusions and forgings in various conditions and sizes, Table 3.0261.

	Mg
5.5	Zn
0.5	Zr

ZK60A

Mg
5.5 Zn
0.5 Zr

ZK60A

- 3.027 Stress concentration.
- 3.0271 Notch properties (see Figure 3.03711).
- 3.0272 Fracture toughness.
- 3.028 Combined properties.
  
- 3.03 Mechanical Properties at Various Temperatures
- 3.031 Tension.
- 3.0311 Stress-strain diagrams.
- 3.03111 Stress-strain curves at room and elevated temperatures for forgings, forged at 600 F from extruded material, Figure 3.03111.
- 3.03112 Stress-strain curves at room and elevated temperatures for forgings, forged at 800 F from cast material, Figure 3.03112.
- 3.03113 Stress-strain curves at room and low temperatures for extrusions, Figure 3.03113.
- 3.0312 Effect of temperature on tensile properties of extrusions, Figure 3.0312.
- 3.0313 Effect of temperature on tensile properties of forgings, Figure 3.0313.
- 3.0314 Effect of temperature on tensile properties of extrusions after various exposure times at test temperature, Figure 3.0314.
- 3.0315 Effect of temperature and strain rate on tensile properties of extrusions in F condition, Figure 3.0315.
- 3.0316 Effect of temperature and strain rate on tensile properties of extrusions in T5 condition, Figure 3.0316.
- 3.032 Compression.
- 3.0321 Stress-strain diagrams.
- 3.0322 Effect of temperature on compressive yield strength of extrusions, Figure 3.0322.
- 3.033 Impact.
- 3.0331 Effect of low temperatures on impact strength of both smooth and notched Charpy bars, Figure 3.0331.
- 3.034 Bending.
- 3.035 Torsion and shear.
- 3.036 Bearing.
- 3.037 Stress concentration.
- 3.0371 Notch properties.
- 3.03711 Effect of low temperatures on notched tensile strength, Figure 3.03711.
- 3.0372 Fracture toughness.
- 3.038 Combined properties.
  
- 3.04 Creep and Creep Rupture Properties
- 3.041 Isochronous stress-strain curves for forgings in T5 condition, Figure 3.041.
- 3.042 Isochronous stress-strain curves for forgings in T6 condition, Figure 3.042.
- 3.043 Total creep strain curves for extrusions, Figure 3.043.
- 3.044 100 hour creep strength of forgings, Table 3.044.
  
- 3.05 Fatigue Properties
- 3.051 Fatigue properties of extrusions, Figure 3.051.
- 3.052 Fatigue properties of forgings, Figure 3.052.
  
- 3.06 Elastic Properties
- 3.061 Poisson's ratio. 0.35 (6).
- 3.062 Modulus of elasticity.
- 3.0621 Effect of temperature on static modulus of elasticity, Figure 3.0621.
- 3.0622 Compressive tangent modulus curves for extrusions of various section sizes in T5 condition, Figure 3.0622.
- 3.0623 Compressive tangent modulus curves for extrusions of various section sizes in F condition, Figure 3.0623.
- 3.0624 Compressive secant modulus curves for extrusions of various section sizes in T5 condition, Figure 3.0624.
- 3.0625 Compressive secant modulus curves for extrusions of various section sizes in F condition, Figure 3.0625.
- 3.063 Modulus of rigidity.  $2.4 \times 10^3$  ksi (9).

4. FABRICATION

- 4.01 Formability
- 4.011 ZK60A, which can be formed by the usual methods and equipment used in the forming of other metals, has good formability at elevated temperatures but relatively poor formability at room temperature.

- 4.012 Formability limits for extrusions, Table 4.012.
- 4.013 Although it is not normally necessary after hot forming, stress relief in accordance with section 1.054 will minimize chances of stress corrosion after severe forming at temperatures appreciably below the maximum recommended forming temperatures.
- 4.014 ZK60A has excellent forgeability in the range 750 to 500 F. Overheating must be avoided to prevent grain growth, which can cause cracking during forging. When reheating during forging is required, it is desirable to reduce the starting temperature by about 50 F for each reheat cycle. A minimum section reduction of 20 percent after the final reheat is recommended to insure a fine grain structure. (23).
  
- 4.02 Machining and Grinding
- 4.021 This alloy, like other magnesium alloys, has exceptionally good machinability, which enables it to be machined at high speeds and feeds. For further details see HZ32A Code 3408, Section 4.021.
- 4.022 The alloy can be chem milled with sulfuric, nitric, or hydrochloric acid of 5 percent strength or greater (24).
  
- 4.03 Welding
- 4.031 Because of a tendency to hot cracking, ZK60A has poor weldability by the gas-shielded arc processes applied to many other magnesium alloys. Although arc welding is not recommended because the results are inconsistent, AZ92A filler metal should be used if welding is necessary. Stress relief in accordance with section 1.054 will minimize the danger of stress corrosion cracking in welded joints. Gas welding is not recommended (17). ZK60A can be satisfactorily resistance welded, spot welding being used to the greatest extent. Spot welds have good static strength (shear and tension), but poor fatigue strength (9)(17).
  
- 4.032
  
- 4.04 Surface Treatment
- For a discussion of various surface treatments that can be applied for corrosion protection see HZ32A, Code 3408, Sections 4.041, 4.042, and 4.043.

TABLE 1.03

Alloy	ZK60A
Source	(1)
Forms	Specifications
Extruded rods, bars, and shapes	AMS 4352 D
	ASTM B107-69
	Federal QQ-M-31b(1) SAE 524
Forgings	AMS 4362 B
	ASTM B91-68
	Federal QQ-M-40b(1) SAE 524
Extruded tubes	AMS 4352 D
	ASTM B107-69
	Federal WW-T-825a SAE 524

TABLE 1.04

Alloy	ZK60A	
Source	(2)(3)(4)(5)	
	Percent	
Element	Minimum	Maximum
Zinc	4.8	6.2
Zirconium	0.45	-
Impurities (total)	-	0.30
Magnesium	Balance	

Mg  
5.5 Zn  
0.5 Zr  
ZK60A

TABLE 1.06

Alloy ZK60A		
Source (9)		
Form	Condition	Brinell Hardness
Extrusions	F	75
	T5	82
Forgings	T5	65
	T6	75

TABLE 1.07

Alloy ZK60A	
Source (9)	
Forms	Condition
Extrusions: rods, bars, solid shapes, tubes, semi-hollow and hollow shapes	F and T5
Forgings: die forgings, hand forgings, and roll forged rings	T5 and T6

TABLE 2.022

Alloy ZK60A			
Condition T5			
Source (10)			
Temperature F	Electrical Conductivity		Electrical Resistivity microhm-in
	Percent LACS	Megmhos per in <sup>3</sup>	
68	30.4	0.447	2.24
100	28.8	0.424	2.36
200	24.6	0.362	2.76
300	21.5	0.316	3.16
400	19.2	0.282	3.55
500	17.4	0.256	3.90

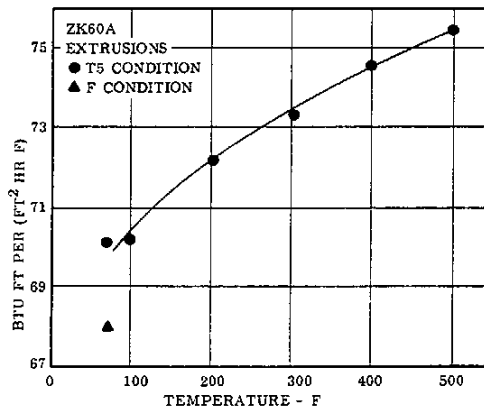


FIG. 2.013 THERMAL CONDUCTIVITY. (10)(11)

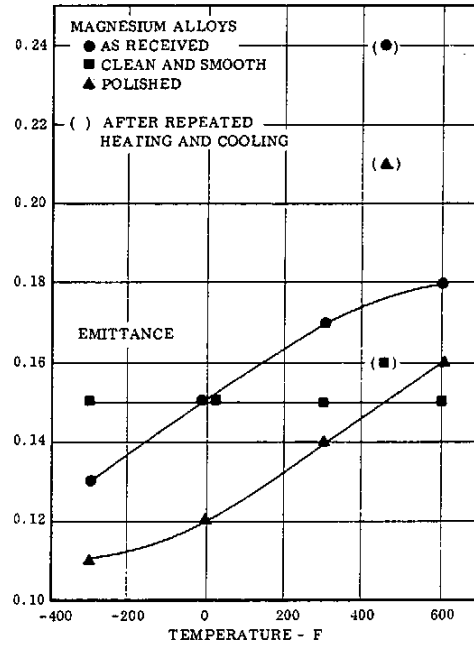


FIG. 2.0241 EFFECT OF TEMPERATURE ON NORMAL TOTAL EMITTANCE OF MAGNESIUM ALLOYS IN GENERAL (11)(12)

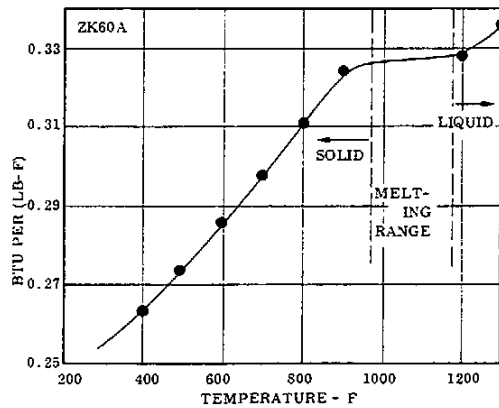


FIG. 2.015 SPECIFIC HEAT. (10)

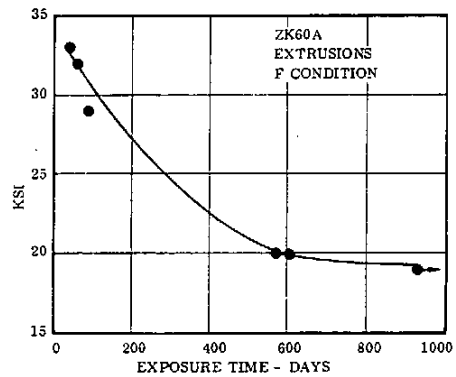


FIG. 2.0312 EFFECT OF STRESS ON TIME TO STRESS CORROSION FAILURE OF SPECIMENS EXPOSED TO ATMOSPHERE IN WASHINGTON D.C. (15)

Mg  
5.5 Zn  
0.5 Zr  
  
ZK60A

TABLE 3.01

Alloy		ZK60A						
Form		Extrusions						
Condition		T5						
Shapes	Source	Cross sectional area - in <sup>2</sup>	Outside diameter in	Section thickness in	F <sub>tu</sub> ksi min	F <sub>ty</sub> ksi min	e (2 in) min	F <sub>ey</sub> ksi min
Bars, rods, and solid shapes	(2)	up to 2	-	-	45.0	36.0	4	30.0
		2 to 3	-	-	45.0	36.0	4	28.0
		3 to 5	-	-	45.0	36.0	4	25.0
		5 to 10	-	-	45.0	34.0	6	23.0
		10 to 25	-	-	45.0	34.0	6	22.0
25 to 40	-	-	43.0	31.0	6	20.0		
Tubing	(2)	up to 3	0.028-0.250	46.0	38.0	4	26.0	
		3 to 8.5	0.094-1.188	44.0	33.0	4	21.0	
Hollow shapes	(2)	all	-	all	46.0	38.0	4	26.0
Bars, rods, and shapes	(5)	up to 5	-	-	45.0	36.0	4	-
Tubing	(5)	up to 3	0.028-0.250	46.0	38.0	4	-	
		3 to 8.5	0.094-1.188	44.0	33.0	4	-	
Hollow shapes	(5)	all	-	all	46.0	38.0	4	-
Form		Extrusions						
Condition		F						
Bars, rods, and shapes	(5)	up to 5	-	-	43.0	31.0	5	-
		5 to 40	-	-	43.0	31.0	4	-
Tubing	(5)	-	up to 3	0.028-0.750	40.0	28.0	5	-
Hollow shapes	(5)	all	-	all	40.0	28.0	5	-
Form		Forgings						
Condition		T5						
Die Forgings	(3)(4)	-	-	up to 3	42.0	26.0	7	-
Hand Forgings	(3)	-	-	up to 6	38.0	20.0	7	-

Note: Test specimens oriented in the direction of maximum flow.

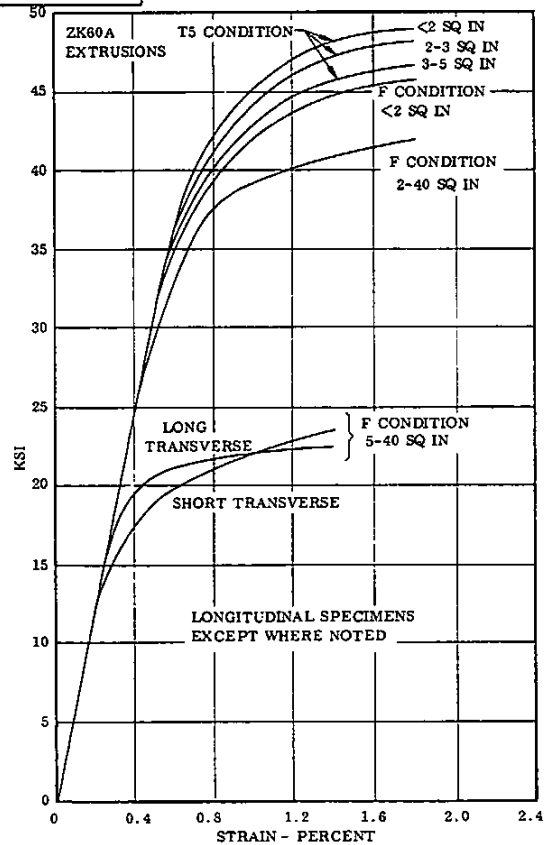


FIG. 3.02111 STRESS-STRAIN CURVES AT ROOM TEMPERATURE FOR EXTRUSIONS IN DIFFERENT CONDITIONS, ORIENTATIONS, AND SECTION SIZES. (9)

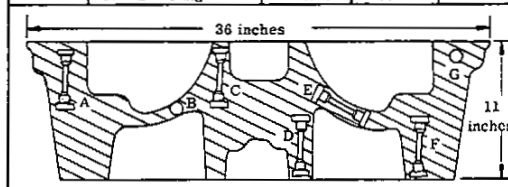
TABLE 3.0212

Alloy		ZK60A				
Source		(9)				
Condition	Form	Cross Sectional Area - sq in	F <sub>TU</sub> ksi	F <sub>TY</sub> ksi	e(2 in)	
T5	Extruded rods, bars, and shapes	under 2.000	53	44	11	
		2.000-2.999	52	43	12	
F	Extruded rods, bars, and shapes	3.000-4.999	51	42	14	
		under 2.000	49	38	14	
		2.000-2.999	49	37	14	
T5	Extruded tubes	3.000-4.999	49	36	14	
		5.000-39.999	48	37	9	
		Dimensions-in				
T5	Extruded tubes	3.0 max OD and 0.028-0.250 wall	50	40	11	
		3.0-8.5 OD and 0.094-1.188 wall	49	39	12	
T5	Extruded hollow shapes	limits not established	50	40	11	
F	Extruded hollow shapes	limits not established	46	34	12	
T5	Forgings	limits not given	44	30	16	
T6	Forgings	limits not given	47	39	11	

Note: All properties apply to longitudinal orientation.

TABLE 3.0214

Alloy		ZK60A			
Form		Aircraft Wheel Forging			
Source		(25)			
Condition	Test Location	F <sub>TU</sub> - ksi	F <sub>TY</sub> - ksi	e(2 in)	
T5	A. Rim Vertical	36.0	19.5	10.5	
	B. Web Tangential	43.5	32.5	18.5	
	C. Center Vertical	39.5	24.0	15.0	
	D. Center Vertical	39.5	23.0	16.5	
	E. Web Radial	41.5	25.0	16.5	
	F. Rim Vertical	42.0	24.0	12.0	
	G. Rim Tangential	45.0	36.5	22.0	
T6	A. Rim Vertical	43.0	34.0	10.5	
	B. Web Tangential	45.5	38.0	11.0	
	C. Center Vertical	43.0	34.0	10.5	
	D. Center Vertical	43.0	34.0	10.5	
	E. Web Radial	42.5	35.0	15.5	
	F. Rim Vertical	43.0	34.0	10.5	
	G. Rim Tangential	45.5	38.0	11.0	



Mg  
5.5 Zn  
0.5 Zr  
ZK60A

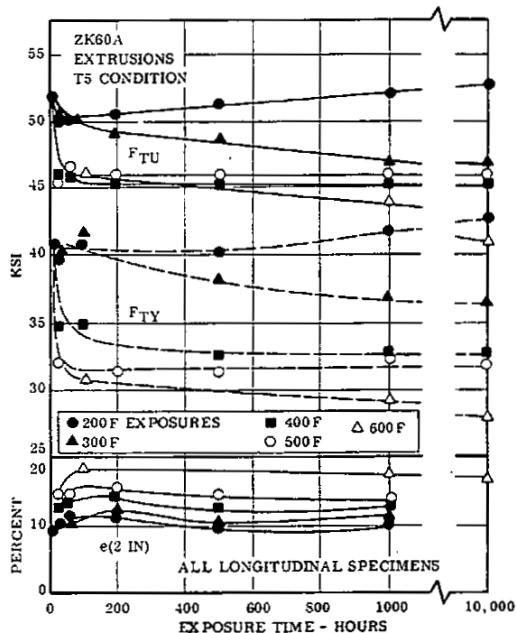


FIG. 3.0213 EFFECT OF EXPOSURES AT ELEVATED TEMPERATURES ON TENSILE PROPERTIES OF EXTRUSIONS AT ROOM TEMPERATURE. (9)(18)

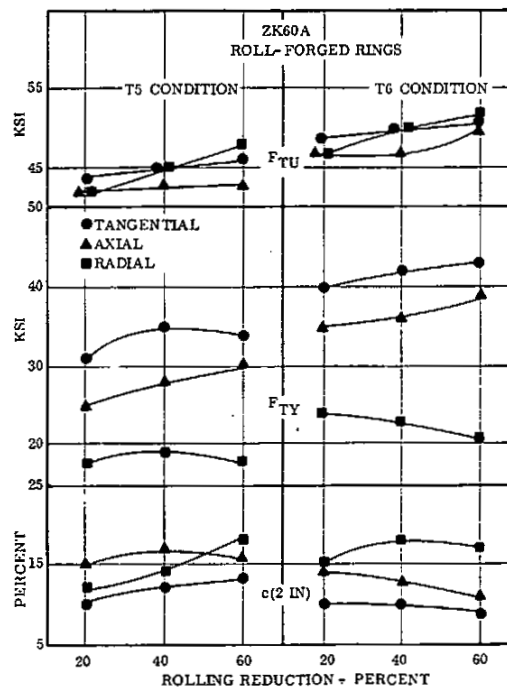


FIG. 3.0215 EFFECT OF ROLLING REDUCTION AND ORIENTATION ON THE TENSILE PROPERTIES OF ROLL-FORGED RINGS PRODUCED DIRECTLY FROM CAST BLANKS. (18)

Mg  
5.5 Zn  
0.5 Zr

ZK60A

TABLE 3.0216

ZK60A						
Form						
Roll Forged Rings						
Source						
(18)						
Condition	Orientation	Rolling Reduction Percent	F <sub>tu</sub> ksi	F <sub>ty</sub> ksi	e (2 in)	
T5	Tangential	40	49	39	13	
		60	51	42	12	
	Axial	40	45	26	14	
		60	45	28	14	
	Radial	40	44	21	20	
		60	45	23	15	
T6	Tangential	40	54	47	10	
		60	51	43	9	
	Axial	40	47	32	11	
		60	46	33	9	
	Radial	40	49	25	22	
		60	50	21	14	

TABLE 3.0222

ZK60A			
Source			
(9)			
Condition	Form	Cross Sectional Area - sq in	F <sub>cy</sub> - ksi
T5	Extruded rods, bars, and shapes	under 2.000	36
		2.000-2.999	31
		3.000-4.999	30
F	Extruded rods, bars, and shapes	under 2.000	33
		2.000-2.999	28
		3.000-4.999	27
		5.000-39.999	23
T5	Extruded tubes	Dimensions - in	
		3.0 max OD and 0.094-1.188 wall	30
T5	Extruded hollow shapes	limits not established	
			29
T5	Forgings	limits not given	
			25

Note: All properties apply to longitudinal orientation.

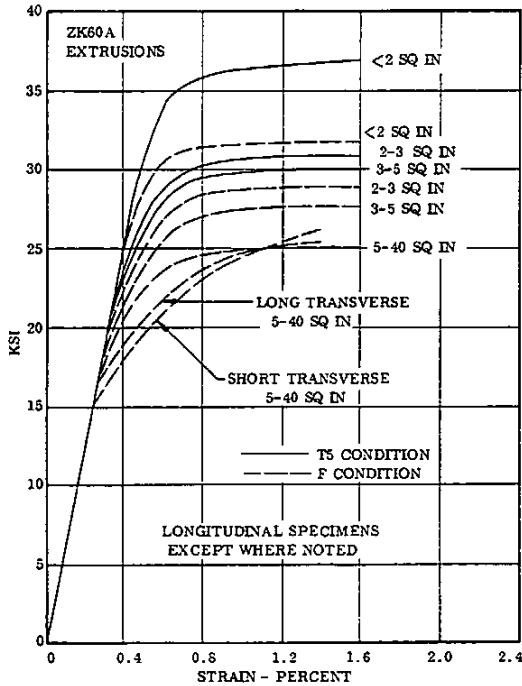


FIG. 3.0221 COMPRESSIVE STRESS-STRAIN CURVES AT ROOM TEMPERATURE FOR EXTRUSIONS IN DIFFERENT CONDITIONS, ORIENTATIONS, AND SECTION SIZES. (9)

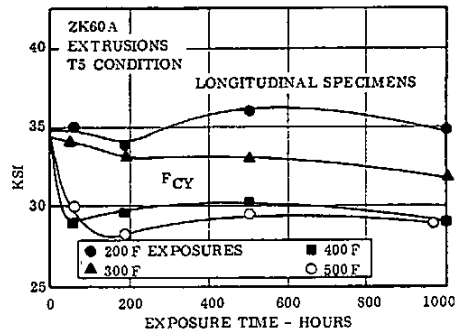


FIG. 3.0223 EFFECT OF EXPOSURES AT ELEVATED TEMPERATURES ON COMPRESSIVE YIELD STRENGTH AT ROOM TEMPERATURE. (9)

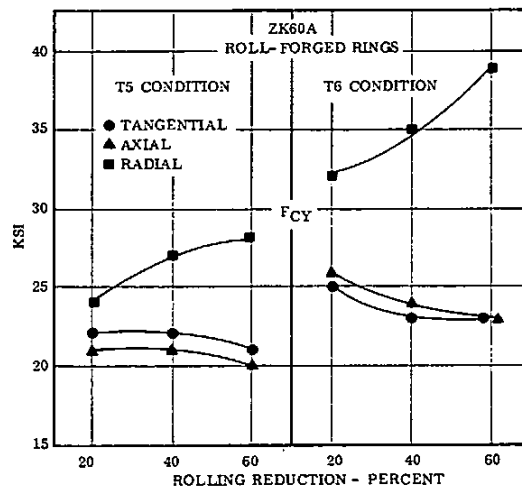
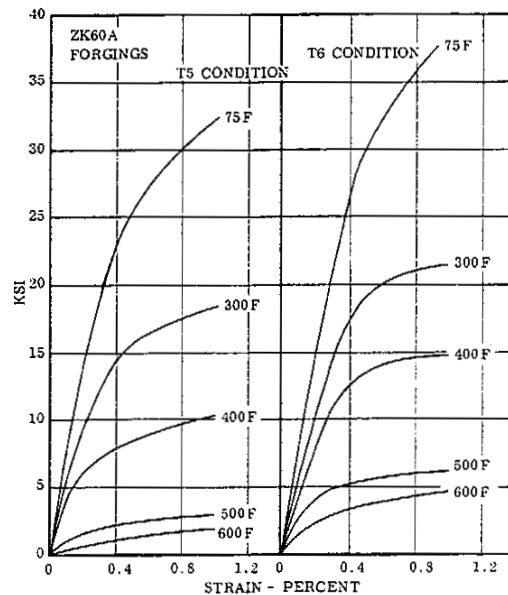


FIG. 3.0224 EFFECT OF ROLLING REDUCTION AND ORIENTATION ON THE COMPRESSIVE YIELD STRENGTH OF ROLL-FORGED RINGS PRODUCED DIRECTLY FROM CAST BLANKS. (15)

TABLE 3.0225

Alloy ZK60A			
Form Roll Forged Rings			
Source (18)			
Condition	Orientation	Rolling Reduction Percent	F <sub>cy</sub> - ksi
T5	Tangential	40	25
		60	27
	Axial	40	22
		60	25
T6	Tangential	40	26
		60	30
	Axial	40	27
		60	22
Radial	40	35	
	60	37	



Mg  
5.5 Zn  
0.5 Zr  
ZK60A

TABLE 3.0251

Alloy ZK60A			
Source (9)			
Condition	Form	Cross Sectional Area - sq in	F <sub>su</sub> - ksi
T5	Extruded rods, bars, and shapes	under 2.000	26
		2.000 - 2.999	26
		3.000 - 4.999	25
F	Extruded rods, bars, and shapes	under 2.000	24
		2.000 - 2.999	24
		3.000 - 4.999	24
		5.000 - 39.999	24
T5	Forgings	limits not given	24
F			25

Note: All properties apply to longitudinal orientation.

FIG. 3.03111 STRESS-STRAIN CURVES AT ROOM AND ELEVATED TEMPERATURES FOR FORGING, FORGED AT 600 F FROM EXTRUDED MATERIAL. (17)

TABLE 3.0261

Alloy ZK60A				
Source (9)				
Condition	Form	Cross sectional area - sq in	F <sub>bru</sub> * ksi	F <sub>bry</sub> * ksi
T5	Extruded rods, bars, and shapes	under 2.000	79	59
		2.000-2.999	75	53
		3.000-4.999	77	52
F	Extruded rods, bars, and shapes	under 2.000	76	56
		2.000-2.999	76	50
		3.000-4.999	76	49
		5.000-39.999	75	44
T5	Forgings	limits not given	55	42
F			65	46

\* edge-distance-to-hole-diameter-ratio of 2.  
Note: All properties apply to longitudinal orientation.

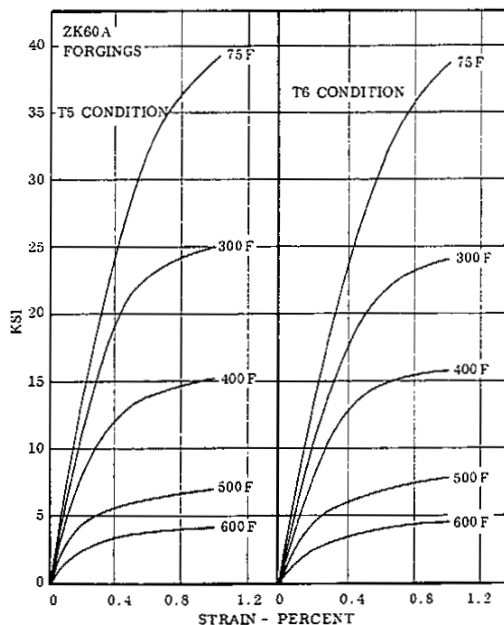


FIG. 3.03112 STRESS-STRAIN CURVES AT ROOM AND ELEVATED TEMPERATURES FOR FORGINGS, FORGED AT 500 F FROM CAST MATERIAL. (17)

Mg
5.5 Zn
0.5 Zr
ZK60A

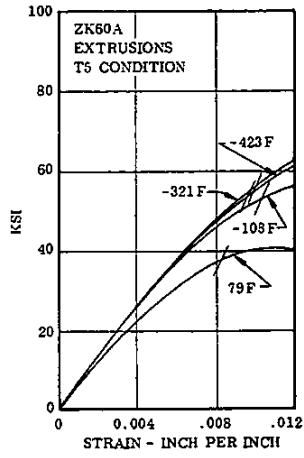


FIG. 3.03113 STRESS-STRAIN CURVES AT ROOM AND LOW TEMPERATURES FOR EXTRUSIONS. (19)

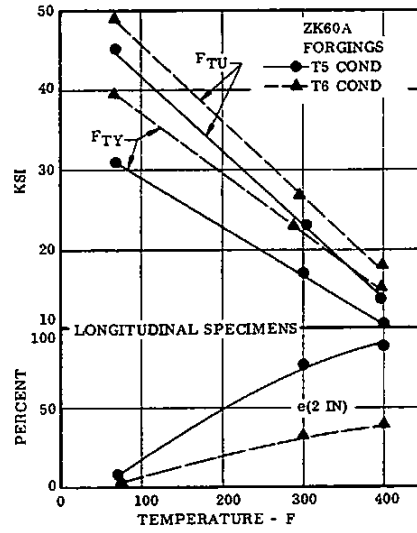


FIG. 3.0313 EFFECT OF TEMPERATURE ON TENSILE PROPERTIES OF FORGINGS. (9)

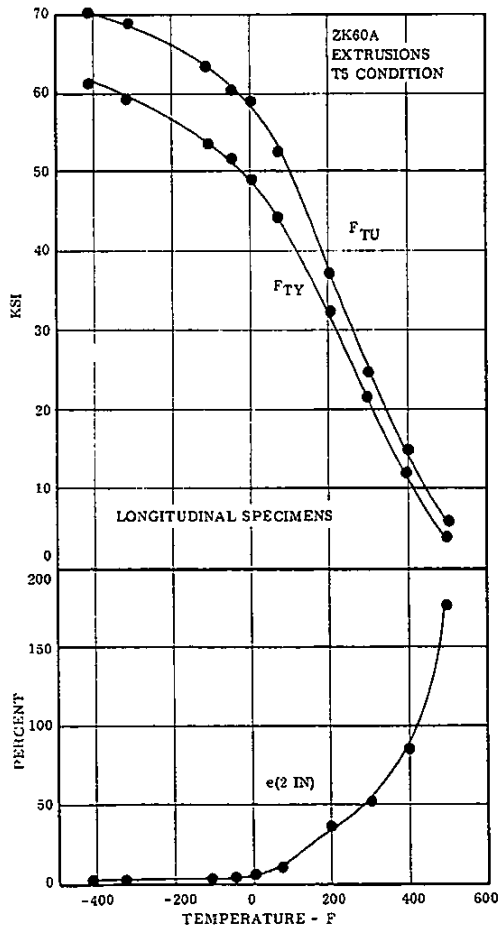


FIG. 3.0312 EFFECT OF TEMPERATURE ON TENSILE PROPERTIES OF EXTRUSIONS. (9)(12)

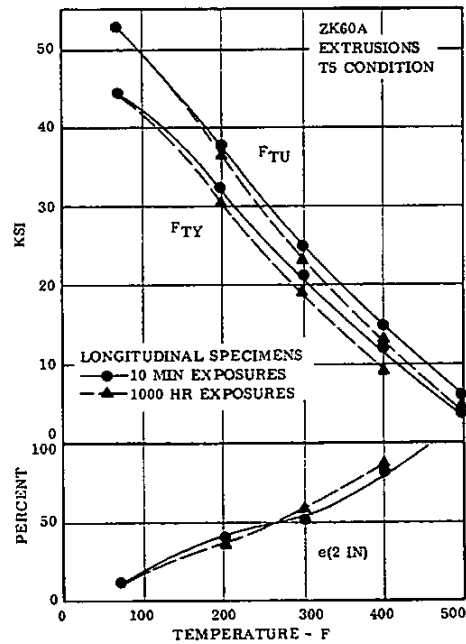


FIG. 3.0314 EFFECT OF TEMPERATURE ON TENSILE PROPERTIES OF EXTRUSIONS AFTER VARIOUS EXPOSURE TIMES AT TEST TEMPERATURE. (9)(20)

Mg
5.5 Zn
0.5 Zr
ZK60A

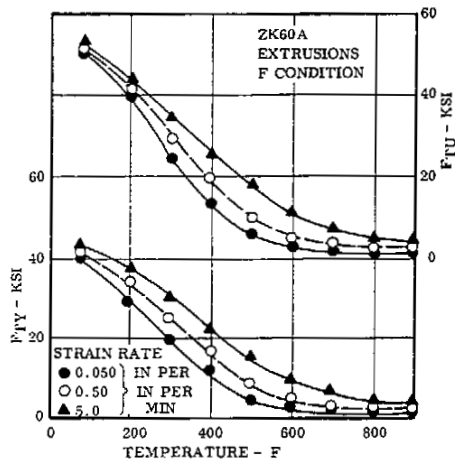


FIG. 3.0315 EFFECT OF TEMPERATURE AND STRAIN RATE ON TENSILE PROPERTIES OF EXTRUSIONS IN F CONDITION. (9)

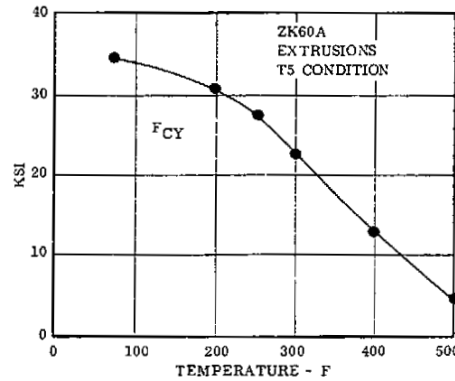


FIG. 3.0322 EFFECT OF TEMPERATURE ON COMPRESSIVE YIELD STRENGTH OF EXTRUSIONS. (9)

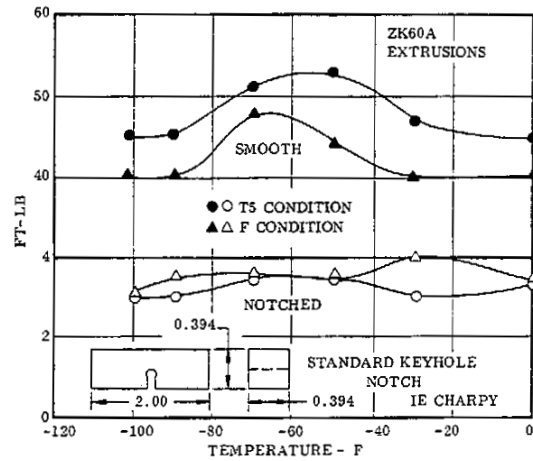


FIG. 3.0331 EFFECTS OF LOW TEMPERATURES ON IMPACT STRENGTH OF BOTH SMOOTH AND NOTCHED CHARPY BARS. (12)

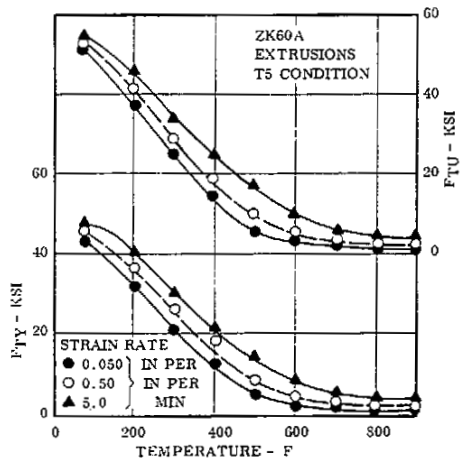


FIG. 3.0316 EFFECT OF TEMPERATURE AND STRAIN RATE ON TENSILE PROPERTIES OF EXTRUSIONS IN T5 CONDITION. (9)

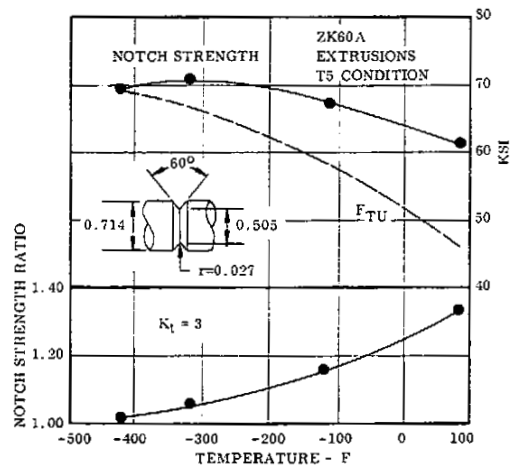


FIG. 3.03711 EFFECT OF LOW TEMPERATURES ON NOTCHED TENSILE STRENGTH. (19)

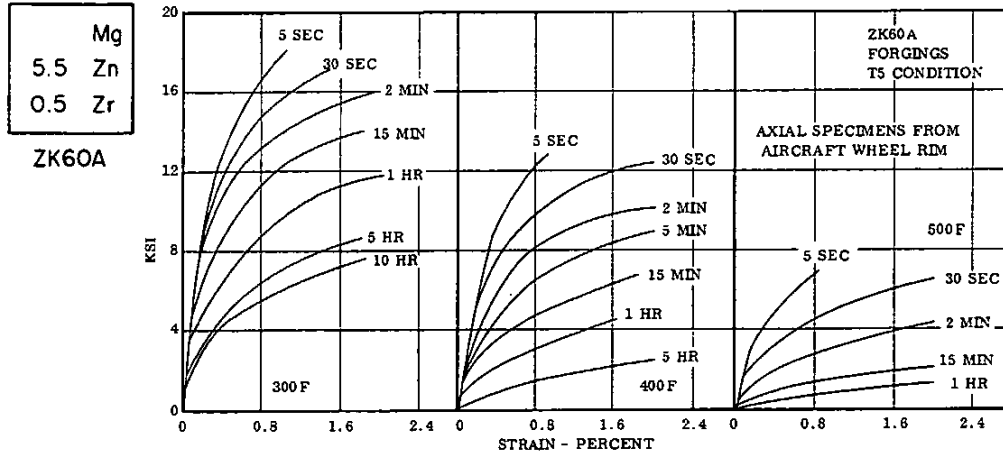


FIG. 3.041 ISOCHRONOUS STRESS-STRAIN CURVES FOR FORGINGS IN T5 CONDITION. (9)

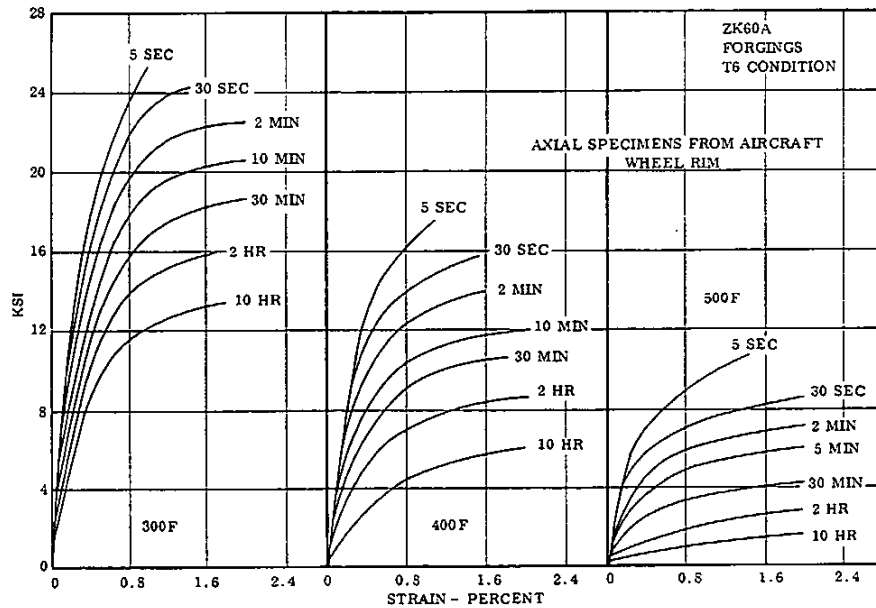


FIG. 3.042 ISOCHRONOUS STRESS-STRAIN CURVES FOR FORGINGS IN T6 CONDITION. (9)

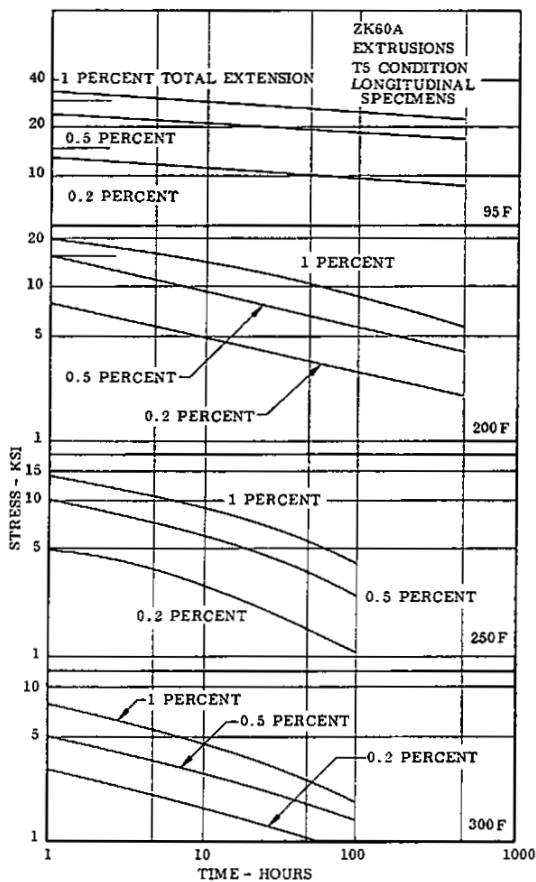


FIG. 3.043 TOTAL CREEP STRAIN CURVES FOR EXTRUSIONS. (9)(20)

TABLE 3.044

Alloy		ZK60A		
Source		(9)		
Condition	Temp	100 hour creep strength - ksi		
		0.1 percent creep extension	0.2 percent total extension	0.5 percent total extension
T5	300	0.9	1.3	2.3
	400	0.2	0.3	0.5
T6	300	3.3	3.8	6.2
	400	1.2	1.5	2.3

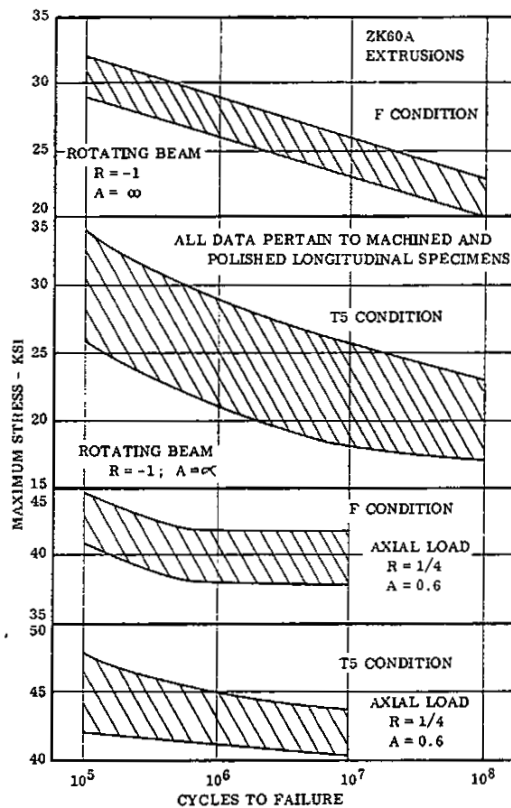


FIG. 3.051 FATIGUE PROPERTIES OF EXTRUSIONS. (9)

Mg  
 5.5 Zn  
 0.5 Zr  
 ZK60A

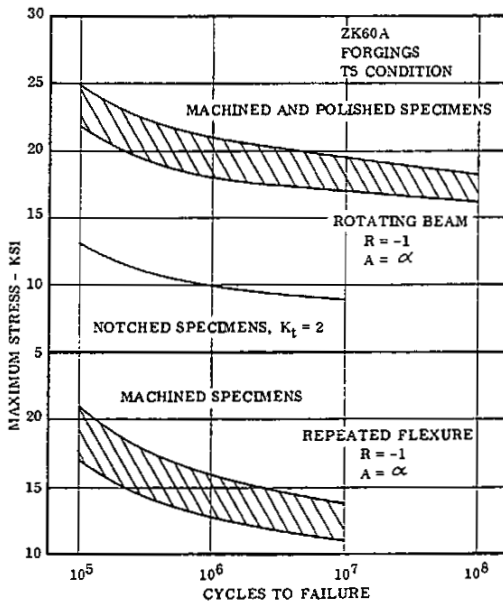


FIG. 3.052 FATIGUE PROPERTIES OF FORGINGS. (9)(21)

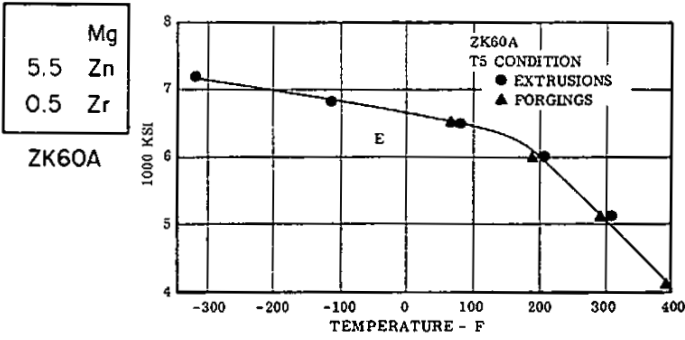


FIG. 3.0621 EFFECT OF TEMPERATURE ON STATIC MODULUS OF ELASTICITY. (9)(19)

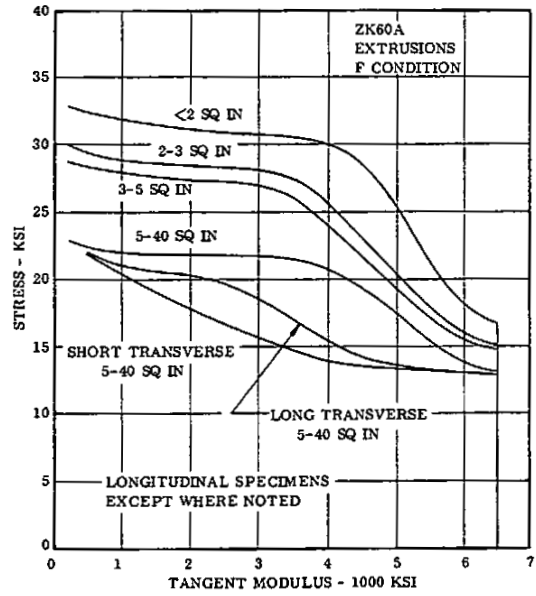


FIG. 3.0623 COMPRESSIVE TANGENT MODULUS CURVES FOR EXTRUSIONS OF VARIOUS SECTION SIZES IN F CONDITION. (9)

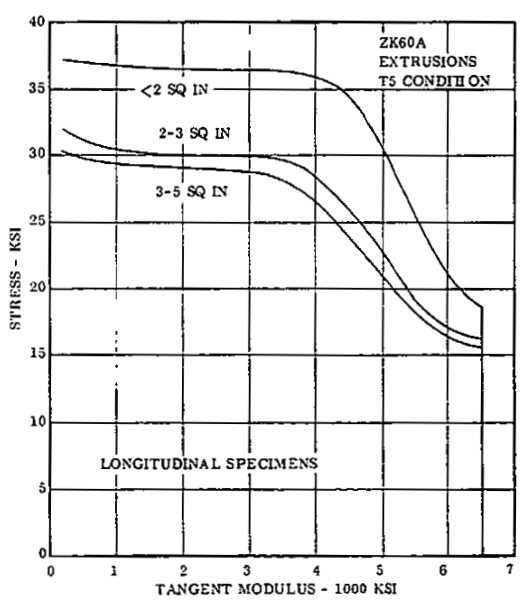


FIG. 3.0622 COMPRESSIVE TANGENT MODULUS CURVES FOR EXTRUSIONS OF VARIOUS SECTION SIZES IN T5 CONDITION. (9)

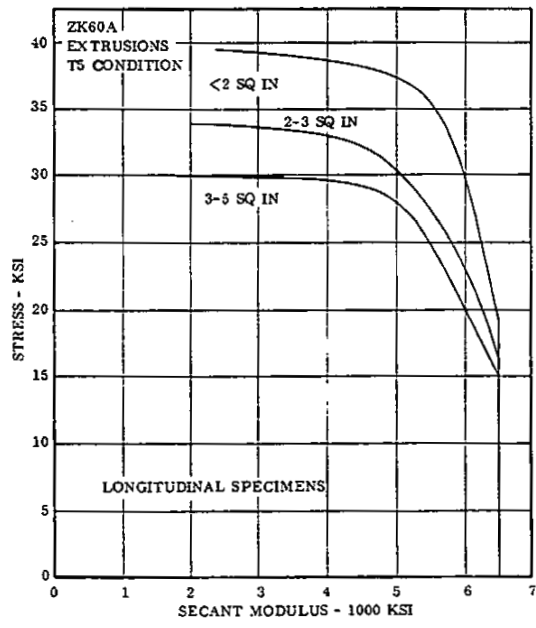


FIG. 3.0624 COMPRESSIVE SECANT MODULUS CURVES FOR EXTRUSIONS OF VARIOUS SECTION SIZES IN T5 CONDITION. (9)

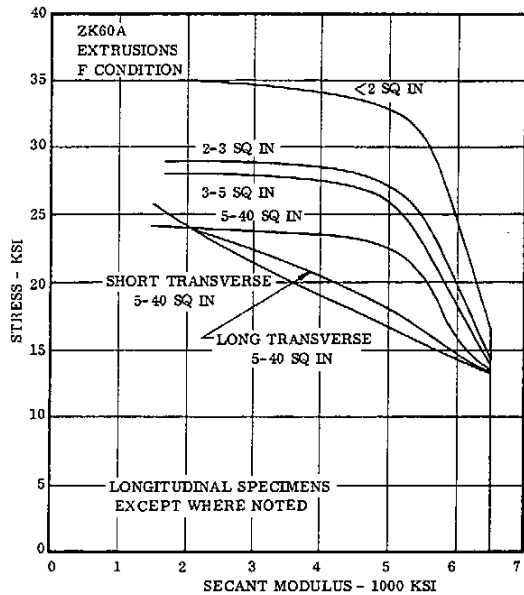


FIG. 3.0625 COMPRESSIVE SECANT MODULUS CURVES FOR EXTRUSIONS OF VARIOUS SECTION SIZES IN F CONDITION. (9)

Mg
5.5 Zn
0.5 Zr
ZK60A

REFERENCES

- "Digest of Specifications for Magnesium Products - 1970" Form No. 141-138-70, The Dow Chemical Company, Midland, Michigan (1970).
- AMS 4352D, Extrusions (May 1, 1968).
- AMS 4362B, Forgings (May 1, 1968).
- ASTM B91-68, Magnesium-Alloy Forgings (1968).
- ASTM B107-69, Magnesium-Alloy Extruded Bars, Rods, Shapes, and Tubes (1969).
- "Heat Treatment of Magnesium Alloys," ASM Metals Handbook, Volume 2, 8th Edition (1964) pp. 292-297.
- "Heat Treating Sand and Permanent Mold Magnesium Castings," Dow Chemical Company, Metal Products Department, Form No. 141-35-68 (1968).
- "Arc Welding Magnesium," Dow Chemical Company, Metal Products Department, Form No. 141-300-67(1967).
- "Magnesium in Design," Dow Chemical Company, Metal Products Department, Form No. 141-213-67 (1967).
- "Physical Properties of Magnesium and Magnesium Alloys," Dow Chemical Company, Metal Products Department, Code 2.51 (April 10, 1967).
- "Heat Transfer Characteristics of Magnesium Alloys," Dow Chemical Company, Metal Products Department, Code 2.46, Appendix A (February 24, 1959).
- "Magnesium Design," Dow Chemical Company, Magnesium Department, Form No. 141-91-457 (1957).
- Jensen, J. W., "Damping Capacity - Causes and Effects" Light Metal Age, Volume 22, pp. 4-8 (December 1964).
- "The Corrosion of Magnesium Alloys," ASM Metals Handbook, Volume 1, 8th Edition, pp. 1086-1094 (1961).
- Logan, H. L., and Hessing, H., "Stress Corrosion of Wrought Magnesium-Base Alloys," Journal of Research of the National Bureau of Standards, Volume 44, pp. 233-243 (March 1950).
- "Mechanical Properties at Various Temperatures of ZK60A-T5 Extrusions," Alcoa Research Laboratories, Technical Data Sheet (August 7, 1957).
- "Properties of Magnesium and Magnesium Alloys," ASM Metals Handbook, Volume 1, 8th Edition, pp. 1095-1112 (1961).
- "Magnesium Rolled Rings," Dow Chemical Company, Metal Products Department, Code 0.4 JFP/HB (December 29, 1964).
- McGee, R. L., Campbell, J. E., Carlson, R. L., and Manning, G. K., "The Mechanical Properties of Certain Aircraft Structural Metals at Very Low Temperatures," Battelle Memorial Institute, WADC TR58-386 (November 1958).
- "Properties of ZK60A and ZK60B Extrusions," Dow Chemical Company, Metal Products Department, Code 0.3HB (April 30, 1959).
- Buckelew, H. C., "Magnesium Alloy Cuts Aircraft Wheel Cost, Weight," SAE Journal, Volume 72, pp. 90-94 (April 1964).
- "Forming Magnesium," Dow Chemical Company, Metal Products Department, Form No. 141-215-64 (1964).
- "Forming Magnesium," Dow Chemical Company, Metal Products Department, Form No. 141-93-55 (1958).
- "Design," booklet published by Magnesium Elektron Ltd., 1270 Avenue of the Americas, New York City 10020.
- "The Selection and Application of Magnesium and Magnesium Alloys," ASM Metals Handbook, Volume 1, 8th Edition, pp. 1067-1086 (1961).

TABLE 4.012

Alloy				
Source				
ZK60A				
(22)				
Form	Condition	Min bend rad at 75 F	Hot forming limits	
			Time and Temp	Min bend rad
Solid extrusions	F	12t	1/2 hr at 550 F	2t
	T5	12t	1/2 hr at 400 F	6.6t
Tubing	F	5D	-	-

Note: t = thickness; D = tube OD

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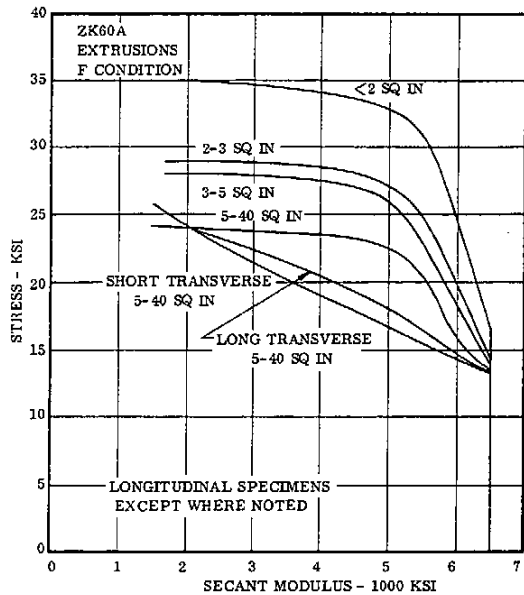


FIG. 3.0625 COMPRESSIVE SECANT MODULUS CURVES FOR EXTRUSIONS OF VARIOUS SECTION SIZES IN F CONDITION. (9)

Mg
5.5 Zn
0.5 Zr
ZK60A

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- "Digest of Specifications for Magnesium Products - 1970" Form No. 141-138-70, The Dow Chemical Company, Midland, Michigan (1970).
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- ASTM B91-68, Magnesium-Alloy Forgings (1968).
- ASTM B107-69, Magnesium-Alloy Extruded Bars, Rods, Shapes, and Tubes (1969).
- "Heat Treatment of Magnesium Alloys," ASM Metals Handbook, Volume 2, 8th Edition (1964) pp. 292-297.
- "Heat Treating Sand and Permanent Mold Magnesium Castings," Dow Chemical Company, Metal Products Department, Form No. 141-35-68 (1968).
- "Arc Welding Magnesium," Dow Chemical Company, Metal Products Department, Form No. 141-300-67(1967).
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- "Heat Transfer Characteristics of Magnesium Alloys," Dow Chemical Company, Metal Products Department, Code 2.46, Appendix A (February 24, 1959).
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- Jensen, J. W., "Damping Capacity - Causes and Effects" Light Metal Age, Volume 22, pp. 4-8 (December 1964).
- "The Corrosion of Magnesium Alloys," ASM Metals Handbook, Volume 1, 8th Edition, pp. 1086-1094 (1961).
- Logan, H. L., and Hessing, H., "Stress Corrosion of Wrought Magnesium-Base Alloys," Journal of Research of the National Bureau of Standards, Volume 44, pp. 233-243 (March 1950).
- "Mechanical Properties at Various Temperatures of ZK60A-T5 Extrusions," Alcoa Research Laboratories, Technical Data Sheet (August 7, 1957).
- "Properties of Magnesium and Magnesium Alloys," ASM Metals Handbook, Volume 1, 8th Edition, pp. 1095-1112 (1961).
- "Magnesium Rolled Rings," Dow Chemical Company, Metal Products Department, Code 0.4 JFP/HB (December 29, 1964).
- McGee, R. L., Campbell, J. E., Carlson, R. L., and Manning, G. K., "The Mechanical Properties of Certain Aircraft Structural Metals at Very Low Temperatures," Battelle Memorial Institute, WADC TR58-386 (November 1958).
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- Buckelew, H. C., "Magnesium Alloy Cuts Aircraft Wheel Cost, Weight," SAE Journal, Volume 72, pp. 90-94 (April 1964).
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Note: t = thickness; D = tube OD

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