



Standard Test Method for Determining Susceptibility to Stress-Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy Products¹

This standard is issued under the fixed designation G47; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ε) indicates an editorial change since the last revision or reapproval.

This standard has been approved for use by agencies of the Department of Defense.

1. Scope

1.1 This test method covers a uniform procedure for characterizing the resistance to stress-corrosion cracking (SCC) of high-strength aluminum alloy wrought products for the guidance of those who perform stress-corrosion tests, for those who prepare stress-corrosion specifications, and for materials engineers.

1.2 This test method covers method of sampling, type of specimen, specimen preparation, test environment, and method of exposure for determining the susceptibility to SCC of 2XXX (with 1.8 to 7.0 % copper) and 7XXX (with 0.4 to 2.8 % copper) aluminum alloy products, particularly when stressed in the short-transverse direction relative to the grain structure.

1.3 The values stated in SI units are to be regarded as standard. The inch-pound units in parentheses are provided for information.

1.4 *This standard does not purport to address all of the safety concerns, if any, associated with its use. It is the responsibility of the user of this standard to establish appropriate safety and health practices and determine the applicability of regulatory limitations prior to use.*

2. Referenced Documents

2.1 ASTM Standards:²

[G38 Practice for Making and Using C-Ring Stress-Corrosion Test Specimens](#)

[G44 Practice for Exposure of Metals and Alloys by Alternate Immersion in Neutral 3.5 % Sodium Chloride Solution](#)

[G49 Practice for Preparation and Use of Direct Tension Stress-Corrosion Test Specimens](#)

[G139 Test Method for Determining Stress-Corrosion Cracking Resistance of Heat-Treatable Aluminum Alloy Products Using Breaking Load Method](#)

3. Summary of Test Method

3.1 This test method provides a comprehensive procedure for accelerated stress-corrosion testing high-strength aluminum alloy product forms, particularly when stressed in the short-transverse grain direction. It specifies tests of constant-strain-loaded, 3.18-mm (0.125-in.) tension specimens or C-rings exposed to 3.5 % sodium chloride (NaCl) solution by alternate immersion, and includes procedures for sampling various manufactured product forms, examination of exposed test specimens, and interpretation of test results.

4. Significance and Use

4.1 The 3.5 % NaCl solution alternate immersion test provides a test environment for detecting materials that would be likely to be susceptible to SCC in natural outdoor environments, especially environments with marine influences.^{3,4,5} For determining actual serviceability of a material, other stress-corrosion tests should be performed in the intended service environment under conditions relating to the end use, including protective measures.

4.2 Although this test method is intended for certain alloy types and for testing products primarily in the short-transverse stressing direction, this method is useful for some other types of alloys and stressing directions.

5. Interferences

5.1 A disadvantage of the 3.5 % NaCl solution alternate immersion test is that severe pitting may develop in the

¹ This test method, which was developed by a joint task group with the Aluminum Association, Inc., is under the jurisdiction of ASTM Committee G01 on Corrosion of Metals and is the direct responsibility of Subcommittee G01.06 on Environmentally Assisted Cracking.

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² For referenced ASTM standards, visit the ASTM website, www.astm.org, or contact ASTM Customer Service at service@astm.org. For *Annual Book of ASTM Standards* volume information, refer to the standard's Document Summary page on the ASTM website.

³ Romans, H. B., *Stress Corrosion Testing*, ASTM STP 425, ASTM, 1967, pp. 182–208.

⁴ Brown, R. H., Sprowls, D. O., and Shumaker, M. B., "The Resistance of Wrought High Strength Aluminum Alloys to Stress Corrosion Cracking," *Stress Corrosion Cracking of Metals—A State of the Art*, ASTM STP 518, ASTM, 1972, pp. 87–118.

⁵ Sprowls, D. O., Summerson, T. J., Ugiansky, G. M., Epstein, S. G., and Craig, H. L., Jr., "Evaluation of a Proposed Standard Method of Testing for Susceptibility to Stress-Corrosion Cracking of High-Strength 7XXX Series Aluminum Alloy Products," *Stress Corrosion-New Approaches*, ASTM STP 610, ASTM, 1976, pp. 3–31.

specimens. Such pitting in tension specimens with relatively small cross section can markedly reduce the effective cross-sectional area and produce a net section stress greater than the nominal gross section stress, resulting in either: (1) fracture by mechanical overload of a material that is not susceptible to SCC; or (2) SCC of a material at an actual stress higher than the intended nominal test stress. The occurrence of either of these phenomena might then interfere with a valid evaluation of materials with relatively high resistance to stress corrosion.

6. Test Specimen

6.1 Type and Size—No single configuration of test specimen is applicable for the many complex shapes and sizes of products that must be evaluated. A tension specimen is preferred because it more consistently provides definite evidence of cracking and should be used whenever the size and shape of the product permits; it also provides a more severe test.

6.1.1 Tension Specimen—The diameter of the reduced section shall be 3.17 ± 0.03 mm (0.125 ± 0.001 in.).

6.1.2 C-Ring (see Practices G38)—The use of C-rings permits short-transverse tests to be made of sections that are too thin or complex for practical tests with a tension specimen. C-rings may be of various sizes as required for the product to be tested, but in no case less than 15.88 ± 0.05 mm (0.625 ± 0.002 in.) in outside diameter. The ratio of diameter to wall thickness shall be kept in the range from 11:1 to 16:1.

6.2 Stressing Direction:

6.2.1 Short-Transverse Tests:

6.2.1.1 For specified material thicknesses of 38.10 mm (1.500 in.) and over, the tension specimen shall be used.

6.2.1.2 For specified material thicknesses of 17.78 through 38.08 mm (0.700 through 1.499 in.), a C-ring shall be used. A tension specimen may be used if consistent with the provisions of Practice G49.

6.2.2 For other stress directions in materials of 6.35 mm (0.250 in.) and over, the tension specimen shall be used.

6.3 Surface Preparation—Test specimens shall be degreased prior to exposure.

7. Sampling and Number of Tests

7.1 Unless otherwise specified, tests shall be performed in the short-transverse direction; the intention is to orient the specimen so that the applied tensile stress is perpendicular to the metal flow lines and in the short-transverse direction relative to the grain structure. In rolled or extruded sections that are approximately round or square, there is no true short-transverse direction because in a transverse plane the grains tend to be equiaxial; and, in such cases, the stress should be directed simply in the transverse direction. If, in certain unusual cases, the grain structure is or tends to be equiaxial also in the longitudinal direction, the stress shall be applied in a direction parallel to the smallest dimension of the product.

7.2 Location of Specimens:

7.2.1 For products stress relieved by stretching (TX51, TX510, TX511, TXX51, TXX510, TXX511), samples shall not be taken from the portion under the stretcher grips.

7.2.2 Rolled Plate—Short-transverse specimens shall be taken so that the region of maximum stress is centered on the mid-plane of the plate and at least $2\frac{1}{2}$ plate thicknesses away

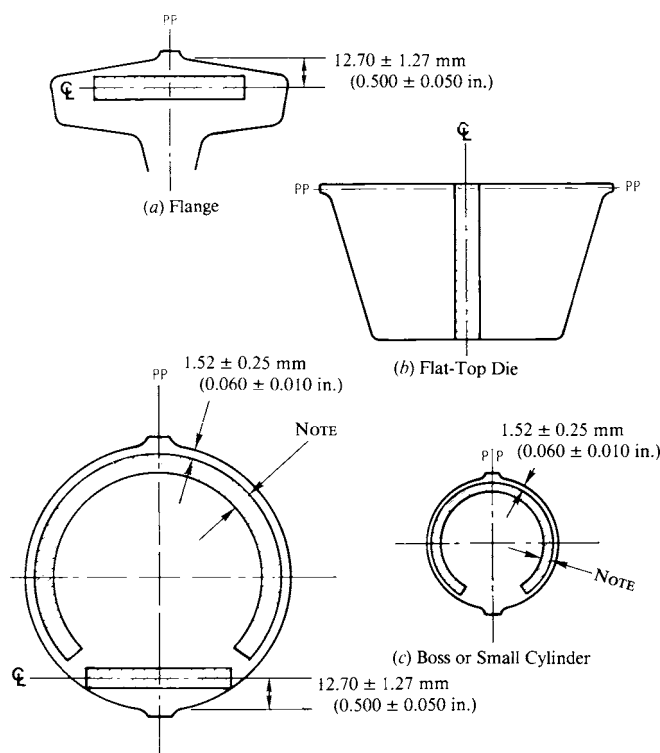
from a side of the plate. (The side of the plate is defined as the edge parallel to the rolling direction.)

7.2.3 Hand Forgings—Short-transverse specimens shall be taken so that the stress is applied in a direction perpendicular to the forging flow lines. The region of maximum stress shall be centered in the forging thickness and approximately on the longitudinal center line of the forging, no less than $\frac{1}{2}$ the section thickness away from “as-heat treated” edges of the forging.

7.2.4 Die Forgings—Because of the wide variety of configurations of die forgings, guidelines are provided for only certain common types of shapes that are widely used. Short-transverse specimens shall be taken so that the stress is applied in a direction perpendicular to the forging flow lines and, if possible, with the region of maximum stress centered on the parting plane. The metal flow pattern in die forgings cannot always be predicted, so only a few general rules are given, and they are illustrated in Fig. 1. Departures from these rules should be made only on the basis of a study of forging flow lines indicating that the intended type of test would not be obtained. In every case, a diagram should be filed with the test results to illustrate specimen locations and orientations.

7.2.4.1 Flanges—The centerline of the specimen shall be 12.70 ± 1.27 mm (0.500 ± 0.050 in.) from the base of the fillet of the flash except for flanges that are too thin, in which case, the specimen should be centered.

7.2.4.2 Flat-Top Die—The tension specimen should be perpendicular to the parting plane and, if possible, centered in the width.



NOTE—Similar to that of typical machined part.

FIG. 1 Recommended Specimen Type and Location for Various Configurations of Die Forgings

7.2.4.3 Boss or Small Cylinder—The C-ring specimen should be centered on the parting plane and with the outside diameter of the ring being 1.52 ± 0.25 mm (0.060 ± 0.010 in.) from the forging surface (see Fig. 1).

7.2.4.4 Large Cylinder—The centerline of tension specimens shall be 12.70 ± 1.27 mm (0.500 ± 0.050 in.) from the base of the flash. If a C-ring is required, its outside diameter shall be 1.52 ± 0.25 mm (0.060 ± 0.010 in.) from the forging surface (see Fig. 1).

7.2.5 Extruded, Rolled, or Cold Finished Rod, Bar, and Shapes:

7.2.5.1 Width-to-Thickness Ratio Greater than 2—Short-transverse specimens shall be taken so that the region of maximum stress is centered in the section thickness, at least one section thickness away from the sides of the product. In the case of complex configurations for which the grain directionality cannot be predicted, specimen location shall be determined by means of macroetched transverse sections to ensure a short-transverse specimen and to avoid regions of nearly equiaxial (transverse) grain flow.

7.2.5.2 Width-to-Thickness Ratio of 2 or Less—Specimens shall be centered in the section thickness so that the region of maximum stress application will be at least one half the section thickness away from a fabricated surface, if possible. These specimens shall be considered to have a “transverse” orientation to the grain structure. When C-rings are required, they shall be taken so that the region of maximum tensile stress is 3.18 ± 0.25 mm (0.125 ± 0.010 in.) from the product surface.

7.3 Number of Specimens—For each sample, which shall be uniform in thickness and grain structure, a minimum of three adjacent replicate specimens shall be tested.

8. Test Environment

8.1 Corrosion Test Environment—Specimens shall be exposed to the alternate 10-min immersion—50-min drying cycle in accordance with Practice G44.

8.2 Length of Exposure—The test duration for 3.18-mm (0.125-in.) tension specimens and C-rings shall be 10 days for 2XXX alloys or 20 days for 7XXX alloys, unless cracking occurs sooner. For specimens to be tested in the long transverse direction, the test duration should be 40 days. Longer nonstandard test durations are likely to cause failures of the 3.18-mm tension specimens as a result of severe pitting as described in 5.1. There shall be no interruptions except as required for periodic inspection of specimens or changing of the solution.

9. Procedure

9.1 Method of Loading:

9.1.1 Tension Specimens—Stress tension specimens in “constant strain”-type fixtures, as in Fig. 3 of Practice G49.

9.1.2 C-rings—Stress C-rings by a method that provides constant strain and produces a tensile stress on the ring outside diameter in accordance with Practice G38.

9.2 Magnitude of Applied Stress—Stress specimens to one or more levels as specified or as required to determine comparative stress corrosion resistance. The application of a stress less than about 103 MPa (15 ksi) is not practicable.

9.3 Examination of Specimens:

9.3.1 Interim Inspection:

Visually inspect specimens each working day for evidence of cracking without removal of corrosion products. Inspection may be facilitated by wetting the specimen with the test solution and by examination at low magnifications.

9.3.2 Final Examination—Perform final examination at a magnification of at least 10X on all surviving specimens after cleaning them in concentrated (70%) nitric acid (HNO_3) at room temperature followed by a water rinse. Section and metallographically examine any C-ring that is considered suspect, as evidenced by linear pitting, to determine whether or not SCC is present. Similar examination of fractured or cracked tension specimens also can be useful to verify SCC as the cause of failure.

10. Interpretation of Results

10.1 Criterion of Failure:

10.1.1 A sample shall be considered to have failed the test if one or more of the specimens fail, except that the retest provisions of Section 11 shall apply.

10.1.2 A specimen that has fractured or which exhibits cracking shall be considered as a stress corrosion failure unless proved otherwise by the provisions of 10.2 and 10.3.

10.2 Macroscopic Examination—Cracking should be clearly differentiated from lined-up pitting. If the presence of SCC is questionable, metallographic examinations should be performed to determine whether or not SCC is present.

NOTE 1—When a specimen fractures within a relatively short time after exposure (ten days or less), metallographic examination is not necessary because such rapid failures are characteristically due to SCC.

10.3 Metallographic Examination:

10.3.1 A specimen that reveals intergranular cracking, even when accompanied by transgranular cracking, shall be considered as an SCC failure. Intergranular fissures that are no deeper than the width of localized areas of intergranular corrosion or, in the case of C-rings, not deeper than those in unstressed or compressively stressed surfaces, shall not be considered as an SCC failure. In the case of tension specimens, the depth of intergranular fissures may be compared to those in an unstressed specimen when available.

10.3.2 A specimen that reveals only pitting corrosion (that is, no intergranular attack), or pitting plus transgranular cracking, shall not be considered as an SCC failure.

NOTE 2—Transgranular cracking in the absence of intergranular attack only occurs in pitted specimens under extremely high stress (intensity) and, for the purpose of this text method, is not considered as a criterion of SCC.

11. Retesting and Resampling

11.1 Retesting shall be permitted only if a single specimen fails by SCC, in which case three replicate specimens shall be tested. If any retest specimen fails, the sample shall be considered to have failed the test.

11.2 If any failure is due to improper preparation of the specimen or to incorrect testing technique, or if the specimen is found to be not representative of the material, the specimen shall be discarded and another specimen substituted.

11.3 When resampling, the required specimens shall be taken from the original sample if possible, or from another sample of the same lot of material.

12. Report

12.1 Report the following information:

12.1.1 Results of all tests, including type and size of specimen, orientation of specimen and number of replicates, stress level, and times to failure.

12.1.2 Identification of alloy, temper, product form, and thickness of materials tested, including reference to applicable specifications.

12.1.3 Any deviation from the procedures outlined above.

13. Precision and Bias

13.1 *Precision:*

13.1.1 The precision of data generated using this test method was evaluated by way of an interlaboratory test program among seven laboratories using aluminum alloy 7075 plate in three tempers; relatively susceptible T651, a more resistant T7X51 (similar to commercial T7651), and highly resistant T7351.

13.1.2 The procedure and raw data are described in detail in ASTM STP 610.⁵ The tests were conducted using five replicate, short transverse specimens tested at various stress levels that were chosen based on the expected performance of the individual tempers. Each of the seven laboratories conducted the test twice so that there were a total 140 specimens in for each combination of temper and stress level. The data have been analyzed with respect to fraction of specimens surviving the standard test period of 20 days. Although three different specimen types were included in the testing, the analysis has been conducted only for 3.18-mm (0.125-in.) diameter tensile bars as described in Practice G49. Reproducibility was evaluated by comparing the seven laboratories against each other, and repeatability was evaluated by comparing the two runs conducted by each laboratory. Treating the three tempers separately, the following conclusions can be drawn.

13.1.2.1 *T651*—Overall, for this susceptible temper, 137 out of 140 specimens failed at stress levels of 103 and 172 MPa (15 and 25 ksi) making calculations of variance not meaningful. Since at least three of five specimens failed in each group and the passing specimens were at two different laboratories, the qualitative observation can be made that all laboratories produced similar results.

13.1.2.2 *T7X51*—This temper with intermediate resistance had a mixture of failing and surviving specimens making the calculations shown in Table 1 and Table 2 meaningful. The calculations show that, depending on the stress level, reproducibility, or laboratory-to-laboratory differences, is responsible for from 60 to 92 % of the variance with repeatability causing the remainder. The variance does depend somewhat on applied stress level as probability of failure would be most consistent at either low stress levels where there are very few, if any failures, or at high stress levels where most, if not all, specimens fail.

13.1.2.3 *T7351*—All specimens of this temper survived at the single stress level of 296 MPa (43 ksi), making calculations of variance not meaningful. However, as with the susceptible T651 temper, the qualitative observation can be made that each laboratory produced the same result.

13.1.3 The results of this interlaboratory test program agree with general experience, which indicates that SCC data will be most consistent under either relatively severe or relatively mild combinations of material and environment. In the first case, the vast majority of specimens will fail quickly, while in the second case, the vast majority of specimens will survive the duration of the test. Variability in results will tend to be highest when material and environment combine to produce a situation of intermediate performance such that some but not all of the specimens fail. The T7X51 material tested in this program fell into the category.

13.1.4 The statement on precision included in previous versions of this test method was based on time-to-failure criteria. That analysis is included in this version as Appendix X1.

13.1.5 Information relevant to the repeatability and reproducibility of the stressing methods and environment called out in this test method can be found in the precision and bias statement of Test Method G139.

13.2 *Bias*—The procedure in Test Method G47 has no bias because the result of the pass-fail stress-corrosion cracking test is defined only in terms of this test method.

14. Keywords

14.1 accelerated testing; aluminum alloys; corrosion; heat-treatable aluminum alloys; stress-corrosion cracking; tension testing

TABLE 1 Calculations of Variance for Fraction of 7075-T7X51 Plate SCC Specimens Surviving in the Interlaboratory Test Program

Applied Stress		Average Fraction Surviving	Repeatability Variance	Reproducibility Variance	Overall Variance
172 MPa	25 ksi	0.886	0.0057	0.0652	0.0709
241 MPa	35 ksi	0.771	0.0743	0.112	0.186
310 MPa	45 ksi	0.557	0.0371	0.0876	0.125

TABLE 2 Calculations of Standard Deviation of 7075-T7X51 Plate SCC Specimens Surviving in the Interlaboratory Test Program

Applied Stress		Average Fraction Surviving	Repeatability Standard Deviation	Reproducibility Standard Deviation	Overall Standard Deviation
172 MPa	25 ksi	0.886	0.0756	0.255	0.266
241 MPa	35 ksi	0.771	0.273	0.335	0.432
310 MPa	45 ksi	0.557	0.193	0.296	0.353

APPENDIX

(Nonmandatory Information)

X1. TIME-TO-FAILURE ANALYSIS OF INTERLABORATORY TEST PROGRAM DATA

X1.1 Previous versions of Test Method G47 used a statement on precision that was based on time-to-failure of specimens. This statement on precision has been changed because the criteria given in the test method for differentiating samples is based on specimens passing or failing (not cracking or cracking) during defined exposure periods. Although the time-to-failure analysis is not directly relevant to results produced by the current procedure, it is included in this appendix because it may provide useful information in certain situations.

X1.2 The graph in Fig. X1.1 shows data from the interlaboratory test program described in STP 610⁵ and Section 13 of this test method. The data in plot show that variability of the data increase as survival time increases. These data agree with the statements made in Section 13 that combinations of material, stress, and environment that produce intermediate levels of resistance will give the highest variability. Combinations of the same factors that produce either low or high resistance will produce less variability because the specimens will either all fail relatively quickly or will not fail at all.

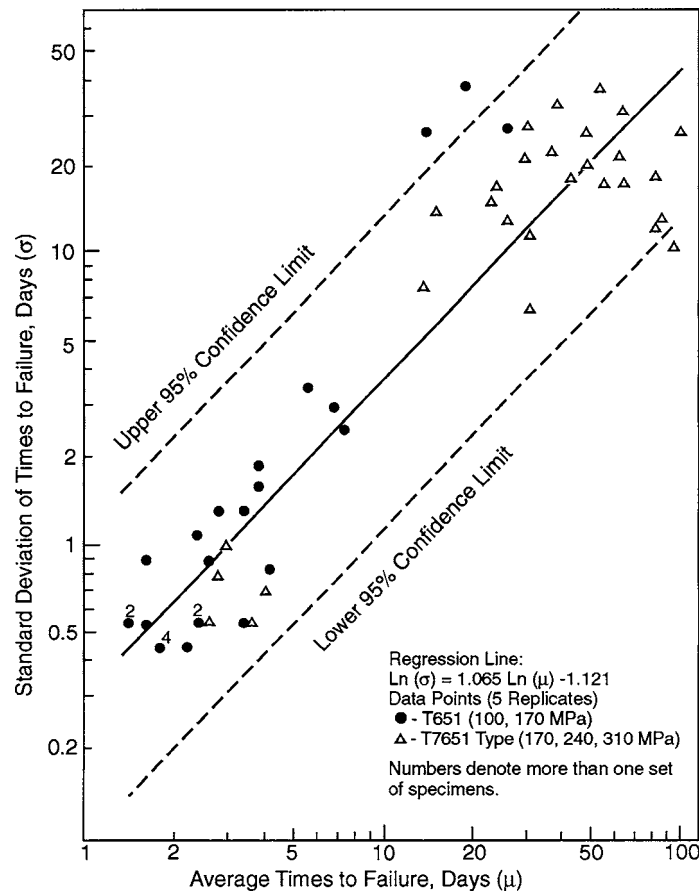


FIG. X1.1 7075 Alloy Short Transverse 3.18 mm Tension Specimens Stressed at Various Levels and Exposed in Quintuplicate to 3.5 % NaCl Solution by Alternate Immersion According to Practice G44

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